Maule Hircraft Orporation
SPENCE AIR BASE: MOULTRY GEORGIA 31/168 -:- PHONE 912/985-2045



10 August 1981 Page 1 of 2

Rev. A dated: 9/3/86

SERVICE LETTER #444

SUBJECT: New Flap Ratchet.

AIRCRAFT AFFECTED: Maule Model M-5-235C, Ser. No. 7001C thru 7320C, 7322C thru 7346C, 7348C thru 7350C, 7352C thru 7362C, 7364C thru 7367C.

COMPLIANCE: Optional.

AUTHORITY: This Service Letter is FAA Approved.

BACKGROUND: Many Maule owners have requested increased flap travel for their airplanes. The new flap ratchet allows 50-80 more travel at each setting. There is also available an alternate lever which is bent upward. This allow more clearance under the lever for easier grasping. Either unit can be used separately.

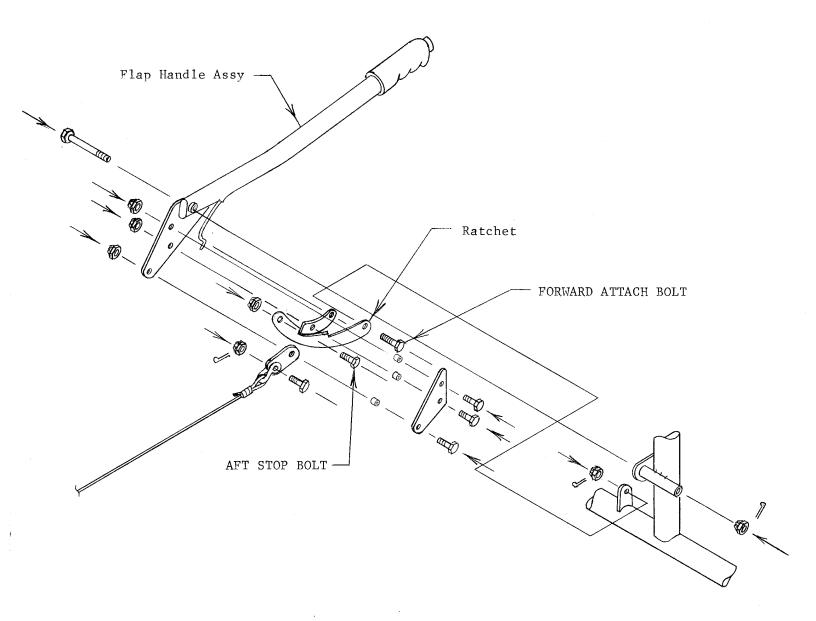
#### MATERIAL REQUIRED:

- (1) Flap Ratchet p/n 3207B
- (1) Alternate Flap Lever Assy p/n 3058F-28 (Opt.)
- (1) Flap Handle Placard (200/400)
- (1) Installation Sketch

#### INSTALLATION INSTRUCTIONS AND ACTION TO BE TAKEN:

- 1. Remove left front seat.
- 2. Remove control cover (ref. Item 6, Group C-4 of Parts Catalog).
- Remove old ratchet: (See Sketch on page 2)
  - (a) Remove forward ratchet attach bolt and nut.
  - (b) Remove aft stop bolt and nut.
  - (c) Slide ratchet forward thru plates of flap handle assembly.
- Install new ratchet:
  - (a) Slide ratchet aft thru plates of flap handle assembly
  - (b) Install forward attach bolt and nut and secure with cotter pin.
  - (c) Install aft stop bolt and nut removed from old ratchet. NOTE: 200/400 ratchet has forward stop bolt and nut deleted.
  - (d) Operational check flap system.
- Reinstall control cover.
- Reinstall left front seat.
- Remove 15°/35° placard from flap handle and apply new placard.
- 8. Remark White and Green arc of Airspeed Indicator in accordance with page 2 of AFM Supplement No. 6. Since the instrument must be opened, it is required that an approved instrument shop do the work.
  NOTE: Remark same if SL #43 is also complied with. (Supp. No. 6 of this SL and Supp. No. 11 of SL #43 have same Airspeed markings.)
- 9. Attach AFM Supplement No. 6 to FAA approved Airplane Flight Manual dated 4/6/76.
- 10. Substitute the enclosed pages 1, 3 and 5 dated 1 Oct 81 of the Weight and Balance data for the corresponding pages you now have.

#### INSTALLATION INSTRUCTIONS



M-5-235C airplanes having ser. no. 7001C thru 7045C, 7047C thru 7052C which have the 20°/40° Flap Ratchet installed in compliance with this Service Letter #44, must also comply with Service Bulletin #7 (AD#86-17-11) concerning fuel crossover line located under the front seats.

When modification is completed, make proper log entries, fill out Service Letter #44 Compliance Record Sheet and return.

# SERVICE LETTER # 44 - COMPLIANCE RECORD

Airplane Serial Number
Airplane Registration Number
The following action was taken with respect to this service letter:
Service Letter #44 completed
Certified by
(Signature)
(Printed name)
Title
(Owner, A&P, IA, etc.)
Date performed
In an effort to keep our mailing list current for sending service bulle- tins, service letter, etc., please fill in the following:
Owner's name
Owner's address
Mail this compliance record to: Maule Air, Inc.
2099 GA Hwy. 133 S.
Moultrie, GA 31788
ATTN: Engineering Records

Maule Hirchard Organian
SPENCE AIR BASI :- MOULTRIE, GEORGIA 31/68 -:- PHONE 912/985-2045



10 August 1981 Page 1 of 2

Rev. A dated: 9/3/86

SERVICE LETTER #44

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AIRCRAFT AFFECTED: Maule Model M-5-235C, Ser. No. 7001C thru 7320C, 7322C thru 7346C, 7348C thru 7350C, 7352C thru 7362C, 7364C thru 7367C.

COMPLIANCE: Optional.

AUTHORITY: This Service Letter is FAA Approved.

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#### MATERIAL REQUIRED:

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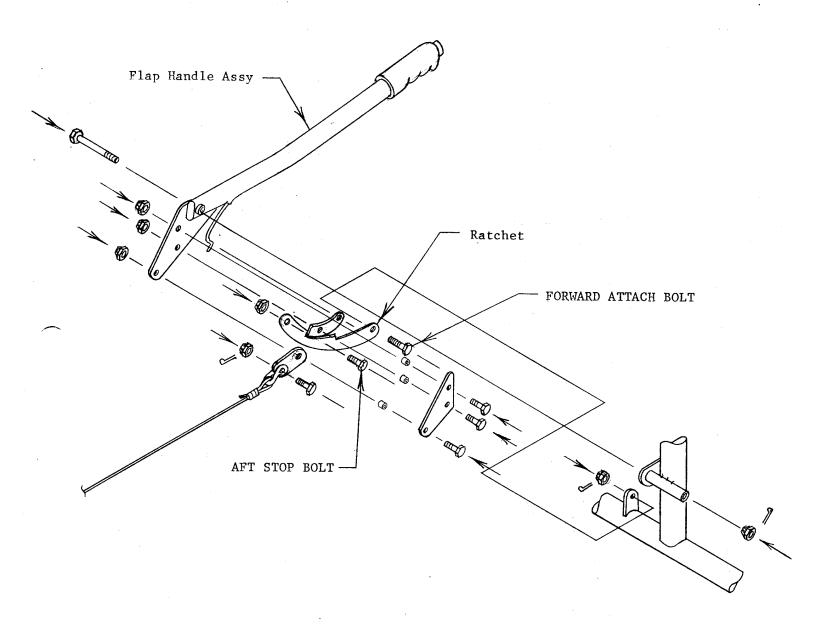
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## INSTALLATION INSTRUCTIONS AND ACTION TO BE TAKEN:

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- 3. Remove old ratchet: (See Sketch on page 2)
  - (a) Remove forward ratchet attach bolt and nut.
  - (b) Remove aft stop bolt and nut.
  - (c) Slide ratchet forward thru plates of flap handle assembly.
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  - (d) Operational check flap system.
- 5. Reinstall control cover.
- 6. Reinstall left front seat.
- 7. Remove  $15^{\circ}/35^{\circ}$  placard from flap handle and apply new placard.
- 8. Remark White and Green arc of Airspeed Indicator in accordance with page 2 of AFM Supplement No. 6. Since the instrument must be opened, it is required that an approved instrument shop do the work.

  NOTE: Remark same if SL #43 is also complied with. (Supp. No. 6 of this SL and Supp. No. 11 of SL #43 have same Airspeed markings.)
- 9. Attach AFM Supplement No. 6 to FAA approved Airplane Flight Manual dated 4/6/76.
- 10. Substitute the enclosed pages 1, 3 and 5 dated 1 Oct 81 of the Weight and Balance data for the corresponding pages you now have.

#### INSTALLATION INSTRUCTIONS



NOTE: M-5-235C airplanes having ser. no. 7001C thru 7045C, 7047C thru 7052C which have the 20°/40° Flap Ratchet installed in compliance with this Service Letter #44, must also comply with Service Bulletin #7 (AD#86-17-11) concerning fuel crossover line located under the front seats.

When modification is completed, make proper log entries, fill out Service Letter #44 Compliance Record Sheet and return.

# SERVICE LETTER # 44 - COMPLIANCE RECORD

Airplane Serial Number
Airplane Registration Number
The following action was taken with respect to this service letter:
Service Letter #44 completed
Certified by
(Signature)
(Printed name)
Title
(Owner, A&P, IA, etc.)
Date performed
In an effort to keep our mailing list current for sending service bulletins, service letter, etc., please fill in the following:
Owner's name
Owner's address
Mail this compliance record to: Maule Aircraft Corporation Engineering Records Spence Field Moultrie, GA 31768

# MAULE AIRCRAFT CORPORATION MOULTRIE, GEORGIA

FAA APPROVED

#### AIRPLANE FLIGHT MANUAL SUPPLEMENT NO. 6

FOR

MODEL M-5-235C

Reg.	No.
Ser.	No.····

This Supplement must be attached to the FAA Approved Airplane Flight Manual dated 6 April 1976 when ratchet p/n 3207B is installed in accordance with Maule Service Letter No. 44.

The information contained herein supersedes and supplements the information of the basic Airplane Flight Manual; for limitations, procedures and performance information not contained in this Supplement, consult the basic Airplane Flight Manual.

FAA APPROVED: Sith & Blight

CHIEF, ENGINEERING AND MANUFACTURING BRANCH FAA, SOUTHERN REGION

DATE: September 10, 1981

#### MOULTRIE, GEORGIA

#### AFM SUPPLEMENT No. 6

M-5-235C

#### OPERATING LIMITATIONS

#### AIRSPEED LIMITS:

AIRSPEED INDICATOR MARKINGS:

Green Arc - 68-145 mph (60-126K)

White Arc -60-94 mph (52-82K)

CENTER OF GRAVITY LIMITS: +11.7 to +20.5 inches @ 1700# or less

#### NORMAL OPERATING PROCEDURES

#### NORMAL FLIGHT OPERATIONS:

#### FLAP SETTING:

Normal Takeoff - 20° (First Notch). No-Flap (0°) takeoff permissible.

Normal Climb - 0°

Best Angle Climb - 20°

Landing - 40° (0° or 20° permissible)

#### CLIMBING:

Best Rate of Climb - 90 mph CAS, no flaps.

Best Angle of Climb - 75 mph CAS, 200 flaps.

FAA APPROVED DATE:

SEP 1 0 1981

PAGE: 1

DATE: 1 Oct 81

# WEIGHT AND BALANCE DATA

AIRCRAFT MODEL M-5-235C

Serial Number, Regist	ration Number				
It is the responsibility of the air					
that the airplane is loaded properly. of gravity and useful load are listed b from the factory. If the airplane has log and/or aircraft records for this in	elow for this airplane as been altered, refer to the	delivered			
WEIGHT AND BALANCE DATA SUMMARY, AS	DELIVERED FROM THE FACTOR	Υ.			
Maximum Gross Weight		lbs.			
Empty Weight		_lbs.			
Useful Load		_lbs.			
Empty Weight Center of Gravit	y	_inches			
Emtpy Weight Moment		_inch lbs.			
NOTE: The empty weight includes 3. and 3 quarts, 6 pounds, of u	**	nuable fuel			
For computing weight and moment for flight, it is more convenient to use a weight and moment that includes the normal engine oil service of 10 quarts of drainable oil at station -34.0 inches for the 0-540 engine and the normal engine oil service of 8 quarts of drainable oil at station -34.0 inches for the IO-540 engine. The weight and moment determined by adding oil service to empt weight and moment will be referred to as <u>basic</u> weight and moment.					
As delivered from the factory	:				
Basic Weight1b.					
Basic Weight Center of Gravityin.					
Basic Weight Momentin. lb.					
CENTER OF GRAVITY RANGE					
At Weight of	Center of Gravity Range				
2300 lbs. 1700 lbs. or less	+12.5 to +20.5 inches +11.7 to +20.5 inches				
NOTE: Straight line variation between given points.					
DATUM: Wing Leading Edge					
CERTIFIED BY	DATE				

PAGE 3

DATE: 1 Oct 81

#### WEIGHT AND BALANCE DATA

5. Calculations for determining weight, c.g. and moment:

a. Center of Gravity (inches) = 
$$(\frac{L \times T}{W})$$
 - D  
ie. C.G. =  $(\frac{X}{W})$  - = inches.

b. Moment (inch pounds) = C.G. x W

ie. Moment = 
$$x$$
 = inch 1b

EXAMPLE OF WEIGHT AND BALANCE CALCULATION FOR LOADED AIRCRAFT.

"An airplane" with basic weight of 1460 lbs. and moment of 15,622 inch lb. is loaded with a pilot and front seat passenger, full main and auxiliary tanks, and 100 pounds of baggage in the baggage compartment.

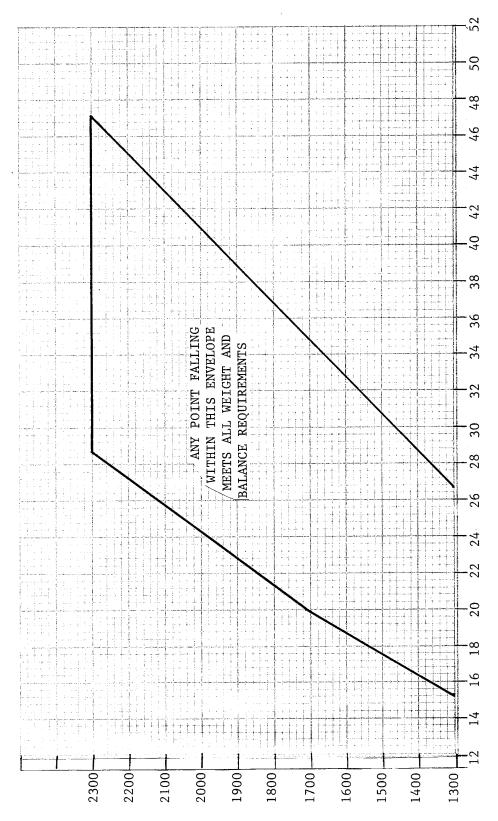
<u>Item</u>	Weight, 1b.	arm, in.	Moment, in. 1b.
Basic Weight Pilot and Front Passenger	1460 340	10.7	15,622 6,800
Fuel - 40 gal. added in Mains plus	378	st.	9.072
23 gal. in Aux's. Baggage (Area "C")	100 ight: 2278	*	7,000
Gross Wei	ight: 2278		38,494

\*Moments can be read directly from the loading graph.

By locating the point corresponding to 2278 lb. aircraft weight and 38,494 inch lb. total moment on the Center of Gravity envelop graph, you can see that this point falls within the envelope, signifying the loading is OK.

TOTAL MOMENT - THOUSANDS OF INCH - POUNDS

CENTER OF GRAVITY ENVELOPE



LOTAL WEIGHT - POUNDS

Form 39